

The strong form of the problem:

$$\operatorname{div} \sigma + b = 0$$

$$u = \bar{u} \text{ on } \Sigma$$

Derivation of a weak form of the problem:

In our case the only one external force is equally distributed body force b per area.

Applying principle of virtual work to the element we obtain equilibrating nodal forces:

$$\iint_{A^e} \delta \epsilon^T \sigma t dA - \iint_{A^e} \delta u^T b t dA = [\delta a^e]^T q^e \quad (1)$$

where $[\delta a^e]^T q^e$ - virtual work by equilibrating nodal forces.

Interpolation of the virtual displacements in terms of nodal values:

$$\delta u^T = [\delta a^e]^T N^T ; \quad \delta \epsilon^T = [\delta a^e]^T B^T$$

Substituting these values into eq. (1) gives:

$$[\delta a^e]^T \left(\iint_{A^e} B^T \sigma t dA - \iint_{A^e} N^T b t dA \right) = [\delta a^e]^T q^e$$

Canceling virtual displacements and substituting stress without consideration of initial strains and stresses:

$$\iint_{A^e} B^T D B t dA \cdot a^e - \iint_{A^e} N^T b t dA = q^e$$

Expression for the global equilibrium:

$$K \cdot a = f$$

where K - global stiffness matrix; f - equilibrating

nodal force

Boundary conditions:

Dirichlet B.C with zero displacements for nodes 1, 2, 3 and with zero horizontal displacement for nodes 6, 5:

$$V_1 = 0$$

$$U_1 = 0$$

$$V_2 = 0$$

$$U_2 = 0$$

$$V_3 = 0$$

$$U_3 = 0$$

$$U_5 = 0$$

$$U_6 = 0$$

Dirichlet B.C with prescribed displacement for node 6:

$$V_6 = -\delta = -0,01 \text{ m}$$

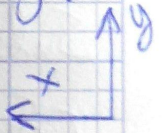
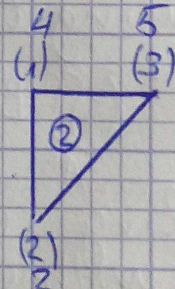
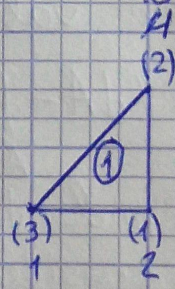
2. Mesh description. Let's introduce local numbering for the elements, starting from the right angle

angle

(1) (2) (3)

1	2	4	1
2	4	2	5
3	3	5	2
4	5	6	4

- connectivity matrix



3. To build linear system of equations, we found local stiffness matrices and assembled into a global one:

$$2 \begin{pmatrix} 1 & 4 & 1 \\ K_{11}^{(1)} & K_{12}^{(1)} & K_{13}^{(1)} \\ K_{21}^{(1)} & K_{22}^{(1)} & K_{23}^{(1)} \\ K_{31}^{(1)} & K_{32}^{(1)} & K_{33}^{(1)} \end{pmatrix}$$

$$3 \begin{pmatrix} 3 & 5 & 2 \\ K_{11}^{(3)} & K_{12}^{(3)} & K_{13}^{(3)} \\ K_{21}^{(3)} & K_{22}^{(3)} & K_{23}^{(3)} \\ K_{31}^{(3)} & K_{32}^{(3)} & K_{33}^{(3)} \end{pmatrix}$$

$$5 \begin{pmatrix} 5 & 6 & 4 \\ K_{11}^{(5)} & K_{12}^{(5)} & K_{13}^{(5)} \\ K_{21}^{(5)} & K_{22}^{(5)} & K_{23}^{(5)} \\ K_{31}^{(5)} & K_{32}^{(5)} & K_{33}^{(5)} \end{pmatrix}$$

$$4 \begin{pmatrix} 4 & 2 & 5 \\ K_{11}^{(2)} & K_{12}^{(2)} & K_{13}^{(2)} \\ K_{21}^{(2)} & K_{22}^{(2)} & K_{23}^{(2)} \\ K_{31}^{(2)} & K_{32}^{(2)} & K_{33}^{(2)} \end{pmatrix}$$

1st element 3rd element 4th element 2nd element

A typical stiffness submatrix element:

$$K_{ij}^{(e)} = \int_{A^e} B_i^T D B_j t dA = \int_{A^e} \frac{1}{2A} e \begin{bmatrix} b_j & 0 & c_j \\ 0 & c_j & b_j \end{bmatrix} \begin{bmatrix} d_{11} & d_{12} & 0 \\ d_{21} & d_{22} & 0 \\ 0 & 0 & d_{33} \end{bmatrix} \frac{1}{2A} e t dA$$

$$\begin{bmatrix} b_j & 0 \\ 0 & c_j \\ c_j & b_j \end{bmatrix} t dA$$

$$K_{ij}^e = \left(\frac{t}{4A} \right)^e \begin{vmatrix} b_i b_j d_{11} + c_i c_j d_{33} & b_i c_j d_{12} + b_j c_i d_{33} \\ c_i c_j d_{21} + b_i b_j d_{33} & b_i b_j d_{33} + c_i c_j d_{22} \end{vmatrix}$$

Linear system of equations $K \cdot a = f$, which is as follows:

$K_{33}^{(1)}$	$K_{31}^{(1)}$	0	$K_{32}^{(1)}$	0	0	0	$Z_1 + f_3^{(1)}$
$K_{31}^{(1)}$	$K_{11}^{(1)} + K_{33}^{(3)} + K_{22}^{(2)}$	$K_{31}^{(3)}$	$K_{12}^{(1)} + K_{21}^{(2)}$	$K_{32}^{(3)} + K_{23}^{(2)}$	0	0	$Z_2 + f_1^{(1)} + f_2^{(2)} + f_3^{(3)}$
0	$K_{31}^{(3)}$	$K_{11}^{(3)}$	0	$K_{12}^{(3)}$	0	0	$Z_3 + f_1^{(3)}$
$K_{32}^{(1)}$	$K_{12}^{(1)} + K_{21}^{(2)}$	0	$K_{22}^{(1)} + K_{33}^{(4)} + K_{11}^{(2)}$	$K_{31}^{(4)} + K_{13}^{(2)}$	$K_{32}^{(4)}$	$U_4 =$	$f_2^{(1)} + f_1^{(2)} + f_3^{(4)}$
0	$K_{32}^{(3)} + K_{23}^{(2)}$	$K_{12}^{(3)}$	$K_{31}^{(4)} + K_{13}^{(2)}$	$K_{22}^{(3)} + K_{11}^{(4)} + K_{33}^{(2)}$	$K_{12}^{(4)}$	0	$f_3^{(2)} + f_2^{(3)} + f_1^{(4)} + Z_5$
0	0	0	$K_{32}^{(4)}$	$K_{12}^{(4)}$	$K_{22}^{(4)}$	0	$f_2^{(4)} + Z_6$

Stiffness matrix consists of 12 rows and 12 columns

After application of boundary conditions the system to be solved has 3 degree of freedom.

4. To calculate FE approximations we firstly compute local stiffness matrices of the elements.

$$K_{11}^{(e)} = \frac{1}{4A} \cdot 10^{10} \begin{vmatrix} 3,28125 & 1,40625 \\ 1,40625 & 3,28125 \end{vmatrix}$$

$$b_1 = 1,5 \quad c_1 = 1,5$$

$$b_2 = 0 \quad c_2 = -1,5$$

$$b_3 = -1,5 \quad c_3 = 0$$

$$d_{11} = d_{22}$$

Using plane stress model:

$$d_{11} = d_{22} = \frac{E}{1-\nu^2} = 1,04166(6) \cdot 10^{10}$$

$$d_{12} = d_{21} = \nu d_{11} = 0,2083(3) \cdot 10^{10}$$

$$d_{33} = \frac{E}{2(1+\nu)} = 0,416(6) \cdot 10^{10}$$

$$K_{12}^{(e)} = \frac{1}{4A} \cdot 10^{10} \begin{vmatrix} -0,9375 \cdot 10^{10} & -0,46875 \cdot 10^{10} \\ -0,9375 \cdot 10^{10} & -2,34375 \cdot 10^{10} \end{vmatrix}$$

$$K_{21}^{(e)} = \frac{1}{4A} \cdot 10^{10} \begin{vmatrix} -0,9375 \cdot 10^{10} & -0,9375 \cdot 10^{10} \\ -0,46875 \cdot 10^{10} & -2,34375 \cdot 10^{10} \end{vmatrix}$$

$$K_{22}^{(e)} = \frac{1}{4A} \cdot 10^{10} \begin{vmatrix} 0,9375 \cdot 10^{10} & 0 \\ 0 & 2,34375 \cdot 10^{10} \end{vmatrix}$$

$$K_{23}^{(e)} = \frac{1}{4A} \cdot 10^{10} \begin{vmatrix} 0 & 0,9375 \cdot 10^{10} \\ 0,46875 \cdot 10^{10} & 0 \end{vmatrix}$$

$$K_{32}^{(e)} = \frac{1}{4A} \begin{vmatrix} 0 & 0,46875 \cdot 10^{10} \\ 0,9375 \cdot 10^{10} & 0 \end{vmatrix}$$

$$K_{33}^{(e)} = \frac{1}{4A} \begin{vmatrix} 2,34375 \cdot 10^{10} & 0 \\ 0 & 0,9375 \cdot 10^{10} \end{vmatrix}$$

$$K_{31}^{(e)} = \frac{1}{4A} \begin{vmatrix} -2,34375 \cdot 10^{10} & -0,46875 \cdot 10^{10} \\ -0,9375 \cdot 10^{10} & -0,9375 \cdot 10^{10} \end{vmatrix}$$

$$K_{13}^{(e)} = \frac{1}{4A} \begin{vmatrix} -2,34375 \cdot 10^{10} & -0,9375 \cdot 10^{10} \\ -0,46875 \cdot 10^{10} & -0,9375 \cdot 10^{10} \end{vmatrix}$$

$$A = \frac{4,5 \cdot 4,5}{2} = 1,125$$

The body force element $f_i^{(e)}$ is equal for each element. As body force equally spaced between 3 nodes of the element, we obtain:

$$f_i^{(e)} = \frac{(A \pm)^e}{3} \begin{vmatrix} b_x \\ b_y \end{vmatrix} = \begin{vmatrix} 0 \\ -375 \end{vmatrix} \quad (b_y = -sg)$$

Eliminating rows and columns from stiffness matrix and moving $K_{32}^{(4)} \delta$; $K_{12}^{(4)} \delta$; $K_{22}^{(4)} \delta$ components to the right hand side we obtained:

$$\begin{vmatrix} K_{22}^{(1)} + K_{33}^{(4)} + K_{11}^{(2)} & K_{31}^{(4)} + K_{13}^{(2)} \\ K_{31}^{(4)} + K_{13}^{(2)} & K_{22}^{(3)} + K_{11}^{(4)} + K_{33}^{(2)} \end{vmatrix} \begin{vmatrix} u_4 \\ v_4 \\ v_5 \end{vmatrix} = \begin{vmatrix} 0 - \delta \cdot \frac{0,46875 \cdot 10^{10}}{4 \cdot 1,125} \\ -1125 - 0 \\ -1125 - \delta \cdot \frac{-2,34375 \cdot 10^{10}}{4 \cdot 1,125} \end{vmatrix}$$

Substituting K_{ij}^e values and solving linear system of 3 equations by inverse matrix method we obtain values of unknowns:

$$u_4 = -0,000128205 \text{ m.}$$

$$v_4 = -0,001132587 \text{ m.}$$

$$v_5 = -0,003867629 \text{ m.}$$